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FAX COVER SHEET FOR MESSAGE FROM:



Kent N. Stone, Patent Attorney NASA Glenn Research Center Mail Stop: 500-118 21000 Brookpark Road Cleveland, Ohio 44135

Voice telephone: (216) 433-8855 FAX: (216) 433-6790

e-mail: k.n.stone@grc.nasa.gov

TO: Ed Look, Director, USPTO Group 3745

DATE: October 18, 2002

MESSAGE

RE: Application Ser. No. 09/498,794

Per our telephone conversation this morning, following is the "Petition to Accept Missing / Item in New Application" and its Attachments, including page 7, which I filed by Express Mail on 4-24-00.

The first page transmitted is the Customer Copy of the U.S. Postal Service Express Mail Label No. EJ596690436US.

V/R

Kent N Stone

Reg. No. 31,883

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LEW 16,833-1: Petition Filed by Expres Mail
4-24-2000

Mailing label No

PATENT

IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

In re Patent Application of Duffy, et al Attorney's Docket No. LEW 16,833-1 Serial No. 09/498,794 Filing Date: 02/01/2000

For: "Self-Tuning Impact Damper for Rotating Blades"

Group Art Unit: 3745

Assistant Commissioner for Patents Attn: Petitions Office Washington D.C. 20231

Box: DAC (Fee)

PETITION TO ACCEPT MISSING ITEM IN NEW APPLICATION UNDER 37 CFR 1.182

Petitioner, the National Aeronautics and Space Administration, an agency of the federal government, respectfully requests that the Commissioner grant this petition to accept one page (page 7) of a specification in an application which was transmitted by Express Mail, but that said page 7 was apparently not received by the PTO.

As more fully set forth in the Declaration, attached, the undersigned certifies that to the best of his knowledge and belief, page 7 of the specification was included in the package of New Application materials sent to the USPTO by Express Mail under mailing label no. EJ770189951US on 2/1/2000.

- B. Pursuant to 37 CFR 1.182, the following are submitted:
- 1. Copy of "Notice of Omitted Item(s) in a Nonprovisional Application"; date mailed: 4/4/2000
 - 2. Declaration of Kent N. Stone (with Attachments, including page 7).
- Charges to the Petitioner's Deposit Account # 14-0116 for the following fees are hereby authorized:

PETITION FEE UNDER 37 CFR 1.17(i).....\$130

The Commissioner is hereby authorized to charge an additional fee which may be required to effect the grant of this petition, or credit any overpayment, to Deposit Account No. 14-0116.

RECEIVED

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OFFICE OF PETITIONS

D. The petitioner's right to prosecute the application as assignee is established by the three assignment documents filed with the original application, for which petitioner has not yet received the PTO's reel and frame numbers.

Respectfully submitted,

Kent N. Stone, Registration No. 31,883

NASA Glenn Research Center 21000 Brookpark Road Mail Stop 500-118 Cleveland, Ohio 44135

telephone: (216) 433-8855

CERTIFICATE OF EXPRESS MAILING UNDER 37 CFR 1.10

I hereby certify that this Petition to Accept Missing Item in a New Application under 37 CFR 1.182 and the documents referred to enclosed therein are being deposited with the United States Postal Service on this date April 27, 2000, in an envelope "Express Mail Post Office to Addressee" under Mailing Label No.

E J 5960 70 436 U.5, addressed to: Assistant Commissioner for Patents, Box: DAC (Fee), Washington D.C. 20231. __, in an envelope as

Kent N. Stone, Registration No. 31,883

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NOV 1 3 2002

OFFICE OF PETITIONS



UNITED STATES DEPARTMENT OF COMMERCE Patent and Trademark Office

Address: COMMISSIONER OF PATENT AND TRADEMARKS Washington, D.C. 20231

APPLICATION NUMBER FILING/RECEIPT DATE

FIRST NAMED APPLICANT

ATTORNEY DOCKET NUMBER

09/498,794

02/01/2000

Kirsten P. Duffy

LEW16,833-1

Kent N Stone NASA Glenn Research Center Mail Stop 500 118 21000 Brookpark Road Cleveland, OH 44135

Date Mailed: 04/04/2000

NOTICE OF OMITTED ITEM(S) IN A NONPROVISIONAL APPLICATION

FILED UNDER 37 CFR 1.53(b)

A filing date has been accorded to the above-identified nonprovisional application papers; however, the following item(s) appear to have been omitted from the application:

- Page(s) page 7 is missing of the specification (description and claims).
- I. Should applicant contend that the above-noted omitted item(s) was in fact deposited in the Patent and Trademark Office (PTO) with the nonprovisional application papers, a copy of this Notice and a petition (and \$130.00 petition fee (37 CFR 1.17(i))) with evidence of such deposit must be filed within TWO MONTHS of date of this Notice. The petition fee will be refunded if is determined that the item(s) was received by the PTO
- II. Should applicant desire to supply the omitted item(s) and accept the date that such omitted item(s) was fill the PTO as the filing date of the above-identified application, a copy of this Notice, the omitted item(s) (with a supplemental oath or declaration in compliance with 37 CFR 1.63 and 1.64 referring to such items), and a punder 37 CFR 1.182 (with the \$130.00 petition fee (37 CFR 1.17(h)) requesting the later filing date must be within TWO MONTHS of the date of this Notice.
- III. The failure to file a petition (and petition fee) under the above options (i) or (ii) within TWO MONTHS of t date of this Notice (37 CFR 1.181(f)) will be treated as a constructive acceptance by the applicant of the application as deposited in the PTO. THIS TWO MONTH PERIOD IS NOT EXTENDABLE UNDER 37 CFR 1.136(a) or (b). In the absence of a timely filed petition in reply to this Notice, the application will maintain a tildate as of the date of deposit of the application papers in the PTO, and original application papers (i.e., the cildisclosure of the invention) will include only those application papers present in the PTO on the date of depo

In the event that applicant elects not to take action pursuant to options (I) or (II) above (thereby constructively electing option (III)), amendment of the specification to renumber the pages consecutively and cancel incomposentences caused by any omitted page(s), and/or amendment of the specification to cancel all references to omitted drawing(s), relabel the drawing figures to be numbered consecutively (if necessary), and correct the references in the specification to the drawing figures to correspond with any relabelled drawing figures, is readily drawing changes should be accompanied by a copy of the drawing figures showing the proposed changered ink. Such amendment and/or correction to the drawing figures, if necessary, should be by way of preliming amendment submitted prior to the first Office action to avoid delays in the prosecution of the application.

to Addressee using Express Mailing label No S 960 90436 CLS on this 2000.

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A copy of this notice MUST be returned with the reply.

Customer Service Center

Initial Patent Examination Division (703) 308-1202

PART 2 - COPY TO BE RETURNED WITH RESPONSE

PATENT

IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

In re Patent Application of Duffy, et al Attorney's Docket No. LEW 16,833-1 Serial No. 09/498,794

Filing Date: 02/01/2000

For: "Self-Tuning Impact Damper for Rotating Blades"

Group Art Unit: 3745

DECLARATION OF KENT N. STONE UNDER 37 CFR 1.68

I, Kent N. Stone, declare as follows:

I have checked my file copy of the documents in utility patent application 09/498,794, and conclude that, to the best of my knowledge and belief, page 7 of the specification to the utility patent application serial no. 09/498,794 was included in the package of new application materials which I sent to the USPTO by Express Mail under mailing label no. EJ770189951US on 2/1/2000. I base this conclusion on the following additional facts:

- 1. I am a patent attorney employed as a civil servant of the United States Government at NASA Glenn Research Center, Cleveland, Ohio.
- 2. The patent function at NASA Glenn Research Center is tasked to the Office of Chief Counsel.
- 3. Since May 1995, I have been the only patent attorney at NASA Glenn Research Center. On average, I file about 12 new applications at the U.S. Patent and Trademark Office (USPTO) per year. The application in the instant case, serial no. 09/498,794 was the fourth application I filed in fiscal year 2000.
- 4. Because of the rather low number of applications I file, and because of the rather modest clerical support the Office of Chief Counsel provides me, I make it a business practice to personally perform all tasks necessary to properly assemble and mail a complete new application package to the USPTO. Hence, for all documents which I intend to include in a new application package, I personally prepare and sign the documents, place stick-on Express Mail labels on the documents, write the mailing label number on the label, prepare a mailing receipt card listing the documents being sent, and photocopy the complete application. Finally, I personally take the complete application package to the Cleveland Airport Post Office for Express Mail service to the USPTO. No clerical assistant is involved in this process.

- 5. I make it a business practice to file all new applications by U.S. Postal Service Express Mail. I always drive the package to the Cleveland Airport Post Office, which is approximately 3 miles from my office at NASA Glenn Research Center, and personally hand it to the postal clerk, who provides me the customer copy of the completed mailing receipt.
- 6. I am absolutely certain that the above procedures were followed with respect to all papers I filed for Serial no. 09/498,794. No one, other than myself, photocopied, assembled, or otherwise touched any papers.
- 7. I distinctly recall the events surrounding the mailing of the application package for serial no. 09/498,794 on February 1, 2000 for the following reasons:
- a) On the late afternoon of February 1, 2000 a severe winter snow storm struck Cleveland, Ohio. At about 7:30 PM, after I had completed assembling the Express Mail package containing the new application documents for serial no. 09/498,794, I took the unusual step of telephoning the Cleveland Airport Post Office, to insure it was open. I was afraid the storm may have prevented the evening shift employees from arriving. I was assured the Post Office was open.
- b) On the late afternoon of February 1, 2000, the photocopy machine on the hallway of the Office of Chief Counsel, Building 500, NASA Glenn Research Center was out of order. I distinctly recall this because the Chief Counsel himself discovered that some non-staff member had damaged the machine trying to remove their personal insurance papers, which had become stuck in the machine. I was forced to use an identical photocopy machine on the second floor of Building 500 to make my file photocopy of the documents I had assembled for the new application package. That photocopy machine worked fine: I do not recall it jamming or skipping any pages. I remember being relieved that it was in good working order.
- 8. I specifically recall that the printer for my office personal computer was near the end of the useful life of its toner cartridge when I printed the execution original of the specification and claims for the application in serial no. 09/498,794. It left a horizontal line near the bottom of each page as it printed out. I notice that the pages of the specification of my file copy of the application contain such a line. Hence, my file copy must have been made from the original, because I recall the original bearing such a line.
- 9. My file copy was made <u>after</u> I had placed the Express Mail labels bearing the mailing label number on the documents. My file copy includes page 7 of the specification. I therefore conclude that the original of page 7 must have been included in the package I sent to the USPTO by Express Mail.
- 10. Attached are True Copies of my file copies of:
 - a) Patent Application Receipt mailing card-showing 29 pages of specification and 12 sheets of drawings were sent to the USPTO; I prepared this card myself.

- b) PTO Form SB/05 Utility Patent Application-showing 29 pages of specification were sent to the USPTO; I prepared this form myself
- c) Application: pages 1-8: page 1 bears the photocopied Express Mail label which I placed upon and signed on the original, and pages 1-8 all show the horizontal line from the near-exhausted toner cartridge that should appear on the original.
- 11. I note that my file copy has a staple through the upper left hand corner which did not entirely pierce all the papers. The 29 pages of specification and 12 sheets of drawings are on the outer limits of what my stapler can reliably go through. I speculate it is possible that in transit, the staple could have worked loose and a page was inadvertently reattached at the USPTO to some other set of documents such as the assignments or the Information Disclosure Statement and prior art references which were transmitted with the specification.
- 12. The page 7 of the specification transmitted herewith is a true copy of the original page 7 which was transmitted with the new application by Express Mail on 2/1/2000.

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine and imprisonment or both under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

Registration No. 31,883

4/24/2000

The attached documents are CERTIFIED TO BE TRUE COPIES of the originals filed on 2/1/2000 in a package under U.S. Postal Service Express Mail, Label No. EJ770189951US:

- Mailing receipt card
- PTO SB/05
- Specification, pages 1-8

Kent N. Stone

Reg. No 31,833

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Patent Application Receipt

NASA CASE NO.

FILING DATE

APPLICATION SERIAL NO. **APPLICANT**

: LEW 16, 833-1

2-1-2000 (TO BE ASSINED)

: Duffy, et a/

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- PTO SB/17 - Fee Transmittel

- NASA FORM 1538 - Declaration, Power of Attny, Patetini

- Assignments of Duffy; Brown; Bayley W/2 Recorbition Forms, each

Application - 29 pages specification + 12 sheets drawing a Into Died. stat w/ 070 1449 & podentil articles

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Stone

Name (PrintType)

Signature

Registration No. (Ammer/Agent)

Deta

LEW: 16,833-1

SELF-TUNING IMPACT DAMPER FOR ROTATING BLADES

ORIGIN OF THE INVENTION

The invention described herein was made in the performance of work under a NASA contract and subjected to the provisions of \$305 of the National Aeronautics and Space Act of 1958, Public Law 85-568(72Stat.435;42U.S.C. 2457).

FIELD OF THE INVENTION

The present invention relates to blades for turbomachinery and, more particularly, to an improved mechanism for dampening vibrations experienced by the blades or disks of turbomachinery so as to extend their operational lives and improve machine performance.

BACKGROUND OF THE INVENTION

Turbomachinery is used in many applications to perform work on or extract work from both gaseous and liquid

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fluids. Examples of such machinery include gas turbines, axial and centrifugal fans, marine and aviation propellers, fan blades, helicopter blades, tail rotors, wind turbines, turbo pumps, and steam and hydraulic power turbines. This machinery may contain one or more broad class of rotating and fixed appendages including blades, vanes, foils, and impellers depending on the need of a particular machine.

Turbines and compressor sections within an axial flow turbine engine, as well as other turbomachinery, generally include a rotator assembly comprising a rotating disk and a plurality of rotor blades circumferentially disposed around the disk. During operation, turbine engine rotator assemblies rotate at a variety of speeds through fluid that varies in temperature, pressure, and density. As a result, the blades may be excited in a number of different modes of vibration. Lower order modes manifest themselves in bending modes and torsion modes, whereas higher order modes have more complex blade deformations.

The rotating blades are subjected to resonating conditions as more fully described in U.S. Patent 5,924,845 ('845) which is herein incorporated by reference. As described in the '845 patent, potentially destructive resonant vibration can occur when the frequency of an alternating excitation force imposed on a-blade is n ar a

natural, or resonant, frequency of the blade. At these resonant frequencies a blade will experience peak vibrations. A typical blade will have more than one resonant frequency, representing the various modes, or ways, in which the blade can vibrate. For example, a jet engine blade may be viewed as a simple cantilever beam. If an alternating excitation force is imposed on the blade at a resonant frequency corresponding to the first mode, it will simply bend back and forth with one wave along its length. If the excitation frequency is at the second mode resonant frequency, the blade will bend with two waves along its length, and so forth. Modern jet engine blades are more like plates than beams, so that the blades have more complicated vibration modes, including, in addition to conventional bending modes, torsion modes and chordwise bending modes. All of these vibration modes combine to determine the actual resonant frequencies for a turbomachine blade.

Excitation forces in turbomachines arise from time dependent fluid flows. For example, one usual source of an excitation force is the aerodynamic force imposed on a rotor blade each time it rotates past a stator vane (a stationary blade). There will be a number of stator vanes past which the rotor blade will travel during one rotor

revolution, causing the frequency of aerodynamic excitation to be equal to the number of stator vanes multiplied by the rotor speed. This is called engine order excitation.

Since there are generally several rows of stator vanes and rotor blades of varying numbers, there are typically many engine order excitation frequencies that a rotor blade will experience. Each of these physical sources of vibratory excitation will cause a different so-called speed line, which is an integer multiple of engine rotor speed.

As first described by W. Campbell in a pioneering 1924 work describing the problems of vibration in turbomachinery, this can be described in a Campbell diagram, where speed lines are plotted on a graph as functions of the rotor speed. Also plotted on the Campbell diagram are the various frequencies at which resonant vibration will occur for each mode of the rotor blade in question. Wherever a speed line, corresponding to a regularly occurring vibratory excitation in a turbomachine, crosses a resonant blade frequency line, resonant blade vibration is possible. This speed line can cross several blade resonant frequencies as the turbomachine speeds up. This means that the excitation frequency on blades will coincide with successively higher resonant frequencies of the blades as rotor speed increases.

Adding one or more dampening devices to a rotor blade will decrease the vibrations and lessen the damage to the blade. In addition, it may allow the turbomachine to be run at a higher speed, improving the efficiency of the machine. Dampening devices may also be added to the rotor disk, which may have vibration modes coupled to the rotor blades. The dampers can then reduce vibrations in the rotor disk and/or the rotor blade.

Rotor blades that are used in axial flow turbines and compressors can be excited to severe levels of vibration when subjected to time-dependent forces as described above. The severe vibrations can damage the material in rotor blades. Devices that absorb and dissipate vibration energy in the blades of rotors in compressors and/or turbines, such as the centrifugal pendulum absorber disclosed in the '845 patent, are known and more of which are disclosed in U.S. Patents 4,182,598; 4,360,088; 4,441,859; 4,484,859; 4,650,167; 5,052,890; 5,232,344; 5,346,362; 5,369,882; 5,498,137; 5,749,705; and 5,820,348 all of which are herein incorporated by reference. It is desired that further improvements for a vibration dampening mechanism for blades be provided.

OBJECTS OF THE INVENTION

It is the primary object of the present invention to provide for a dampening mechanism that absorbs and dissipates vibration energy in the blades or disks of compressors and/or turbines so to extend the service life thereof.

It is another object of the present invention to provide for a dampening mechanism that uses the technique of self-tuned impact dampening.

It is another object of the present invention to provide for a self-tuned impact dampening mechanism that is adjustable to dampen engine-order resonant vibrations along a speed line.

It is a further object of the present invention to provide for a self-tuning dampening mechanism that utilizes a ball-in-spherical trough configuration.

It is a still further object of the present invention to provide for a self-tuning dampening mechanism that can be located internal or external to the blade for which it provides the dampening features so as to extend its operational life.

SUMMARY OF THE INVENTION

This invention is directed to a self-tuning dampening mechanism for a rotating appendage for a turbomachine which absorbs and dissipates the vibration energy in the appendage so as to extend its operational life.

The vibration damper device comprises a member coupled to a rotor having a frequency of vibration with the member being subjected to fluid-flow forces when the rotor is rotated. The member has one or more cavities with walls which confine a rattling mass in each cavity having parameters that are selected in accordance with the anticipated frequency of aerodynamic excitation associated with a speed line on the appropriate Campbell diagram.

In one embodiment, the vibration damper device has a trough configuration having a spherical bottom and a rattling mass, such as a ball, having a resonant frequency that corresponds to the anticipated frequency of aerodynamic excitation.

BRIEF DESCRIPTION OF THE DRAWINGS

For a better understanding of the nature and objects of the present invention, reference should be made to the

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following detailed description taken in conjunction with the accompanied drawings, in which like parts are given like reference numbers, and wherein:

Fig. 1 is a schematic drawing of a turbine wheel with blades and a shaft;

Fig. 2 is composed of Figs 2(A), 2(B), and 2(C), wherein Figs 2(A), and 2(B) schematically illustrate prior art dampening mechanisms, and Fig. 2(C) schematically illustrates the self-tuning impact damper mechanism of the present invention;

Fig. 3 illustrates a Campbell diagram showing speed lines and resonant frequencies plotted against rotor speed;

Fig. 4 illustrates a two (2) degree-of-freedom system;

Fig. 5 illustrates various curves associated with tuned mass amplification factor calculations;

Fig. 6 is composed of Fig. 6(A) and 6(B) that respectively illustrate the amplification factor calculation curves and damping factor calculation curves associated with the self-tuning impact damper of the present invention;

Fig. 7 illustrates response curves associated with the tuned mass damper effectiveness;